



# IBEX 2006 Session 101

## Design Basics for Composite Structures

Jim Antrim – Antrim Associates

Alex Kozloff – Kozloff Enterprises

Jeremy Laundergan – Askeland Engineering

# What are composites?

## General term:

Any combination of materials bonded together to act as one engineering unit, including, for example:

- METALS
- WOOD
- FOAM
- PLASTIC
- F.R.P
- Where your imagination goes....

## Specific term:

(As used in this talk)

F.R.P. –

FIBER reinforced PLASTICS

F.R.P. usually used as solid *laminate* or as laminated “sandwich”

# Familiar Materials vs. Composites

- METALS

Isotropic- Uniform properties in all directions

Yield (permanent bend) before failure

- WOOD

Nature's unidirectional: has a grain, and is much stronger in line with grain

- PLYWOOD

*Laminated* wood, bi-directional

- PLASTICS

Homogenous

Usually relatively weak

Soft or hard

Non-linear stress/strain

## COMPOSITE LAMINATE

- Chopped strand mat

Quasi isotropic

- Other composite fabrics are *directional*

One-directional rovings or fibers combined with weaving, stitching, and/or layering

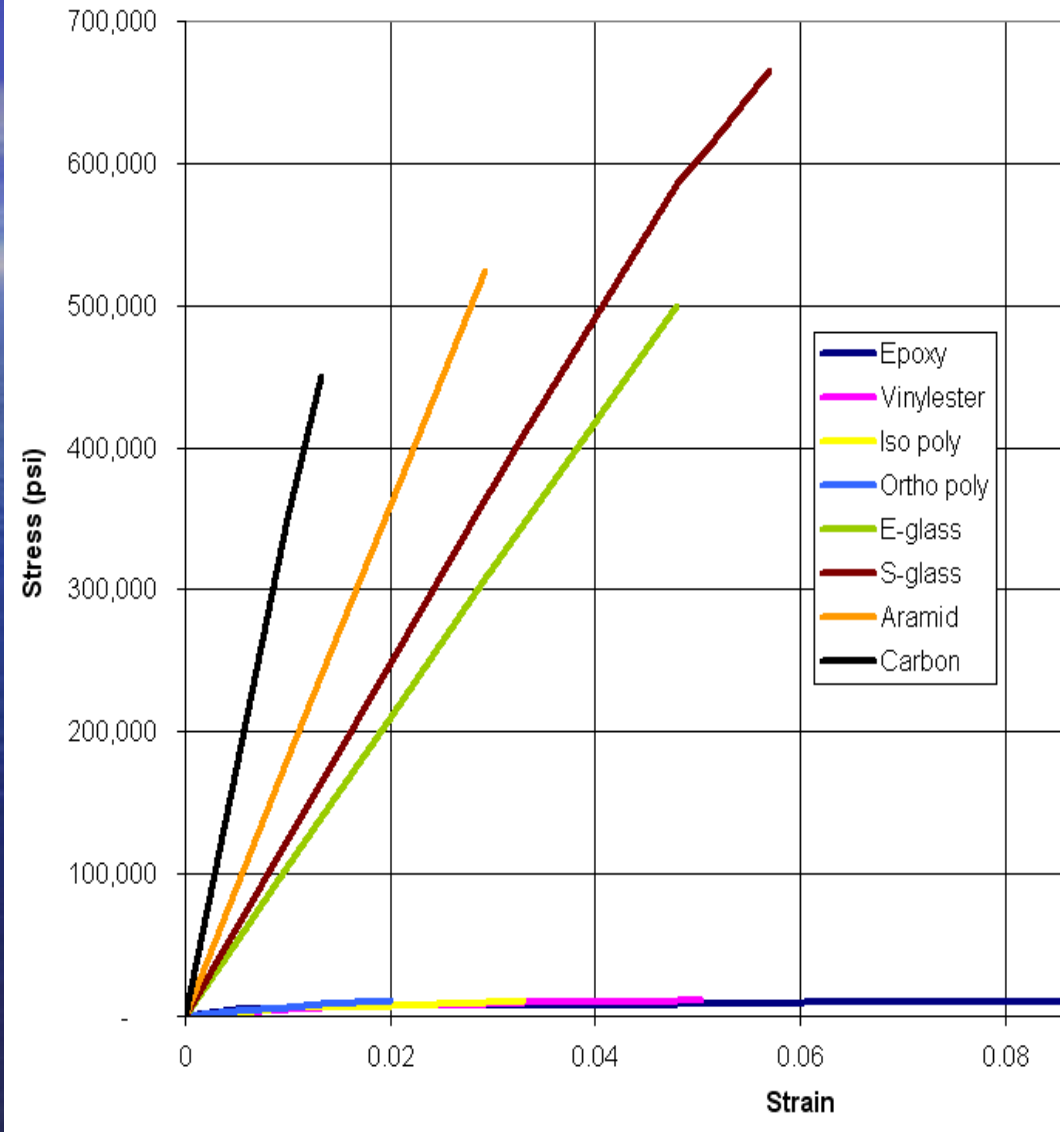
Think of the fiber direction as *grain* (as in wood)

It is both FIBER REINFORCED PLASTIC and PLASTIC BONDED FIBERS

# Stress/Strain for Fibers & Resin.

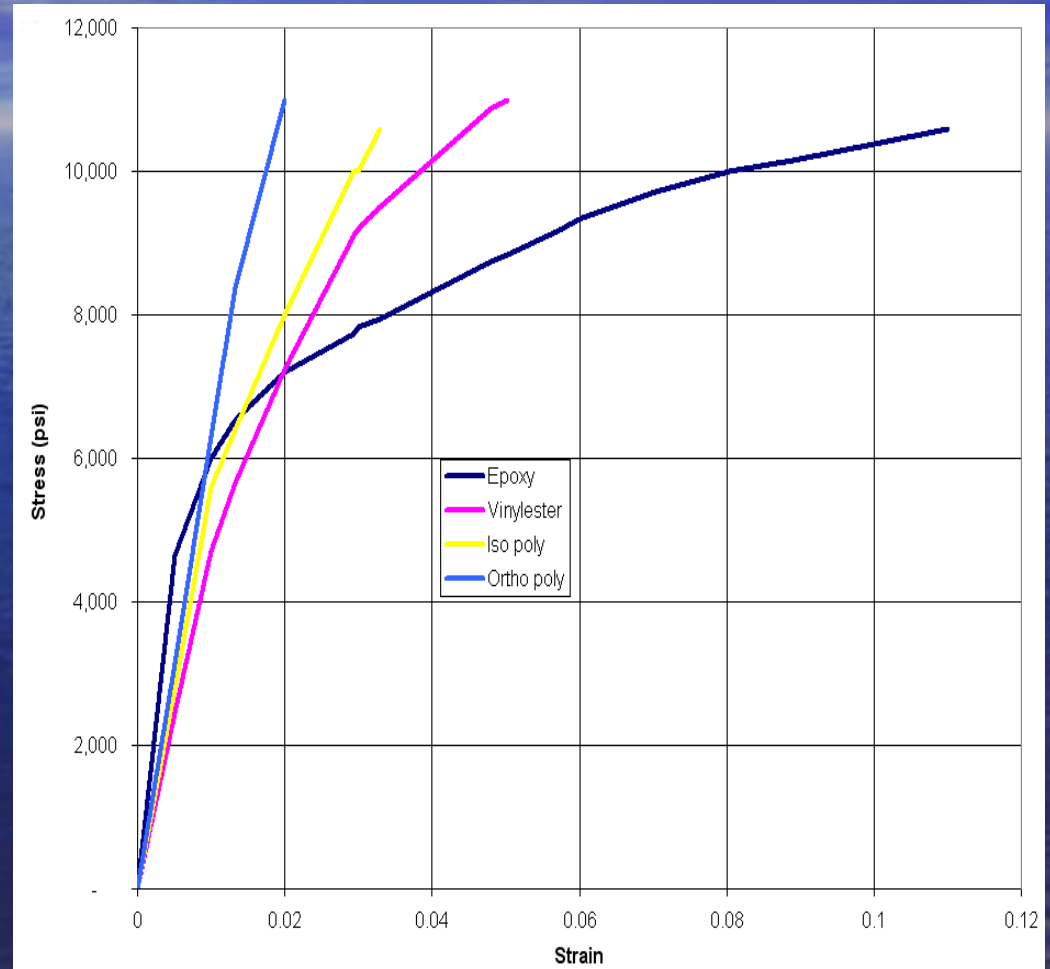
(Isolated, not in a laminate)

- Fibers do not yield
- Fibers are vastly stronger than resin. This is why “grain” direction is important
- Stiff fibers (e.g. carbon) carry high load at low strain (low stretch)
- Lower quality resins fail before fibers – fatigue & laminate strength limiting
- High quality resin have high strain. Fatigue is greatly reduced. Fibers can reach load capacity



# Stress/Strain for various resin types

- Same plot as previous
- These are generic. Properties can vary significantly
- Ortho polyesters are typically stiff/more brittle. Strain to failure less than most fibers
- Vinylesters and Epoxies have much higher strain to failure, therefore less fatigue in a laminate
- More fiber strength is available when resin matrix has higher strain at failure



# Use composites effectively

- Align fibers (“grain”) with the load
- Even when load path is well known, 100% unidirectional laminate is rarely wise.

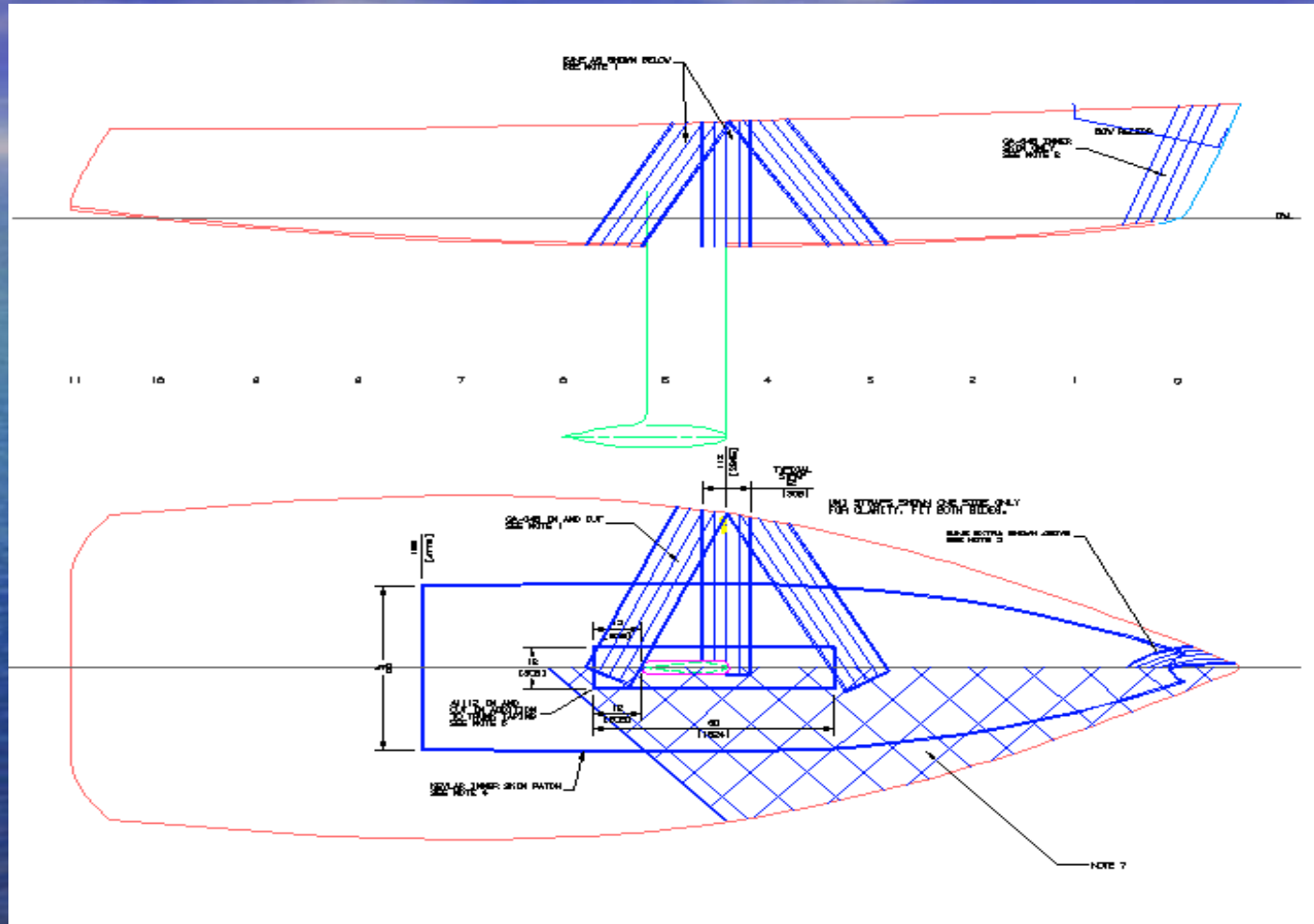
(Due to Poisson effects, at least 20% of fibers off axis is usually desirable when load path is clearly defined.)

- For panels, primary grain should run in the short direction across framing.

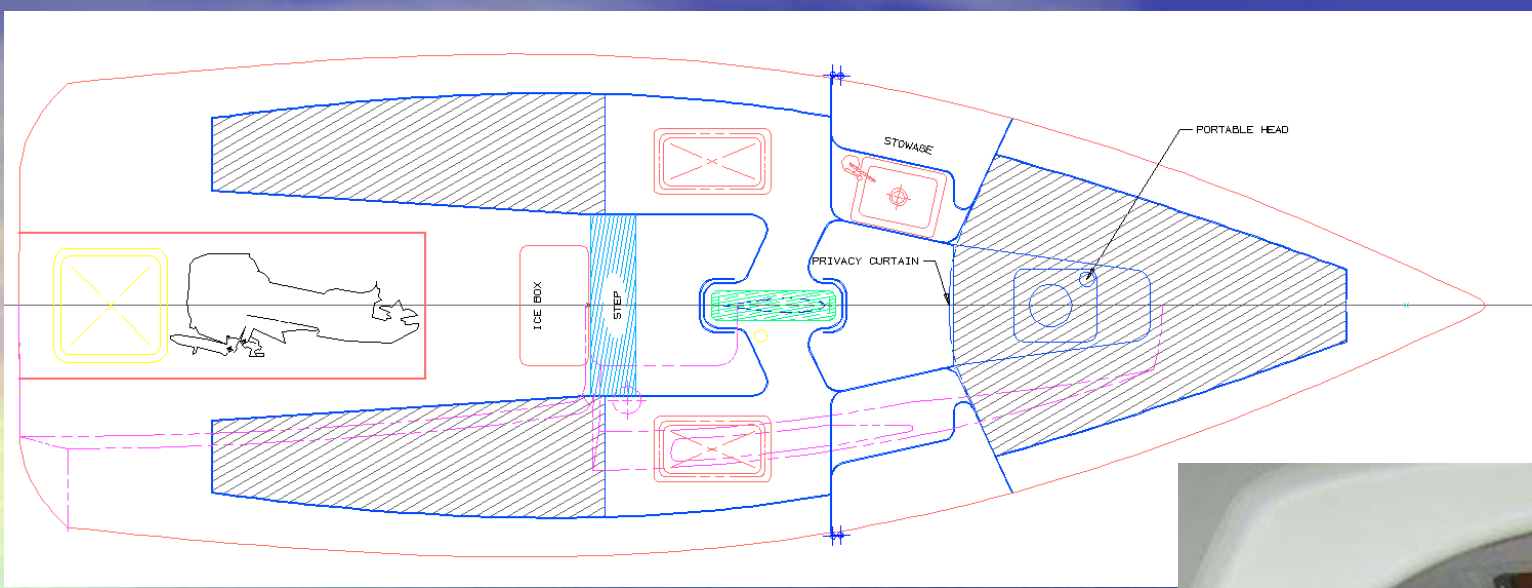
(For a wood floor, would you lay your planking across the frames or parallel to them?)

- Exotic, expensive material where it is most effective

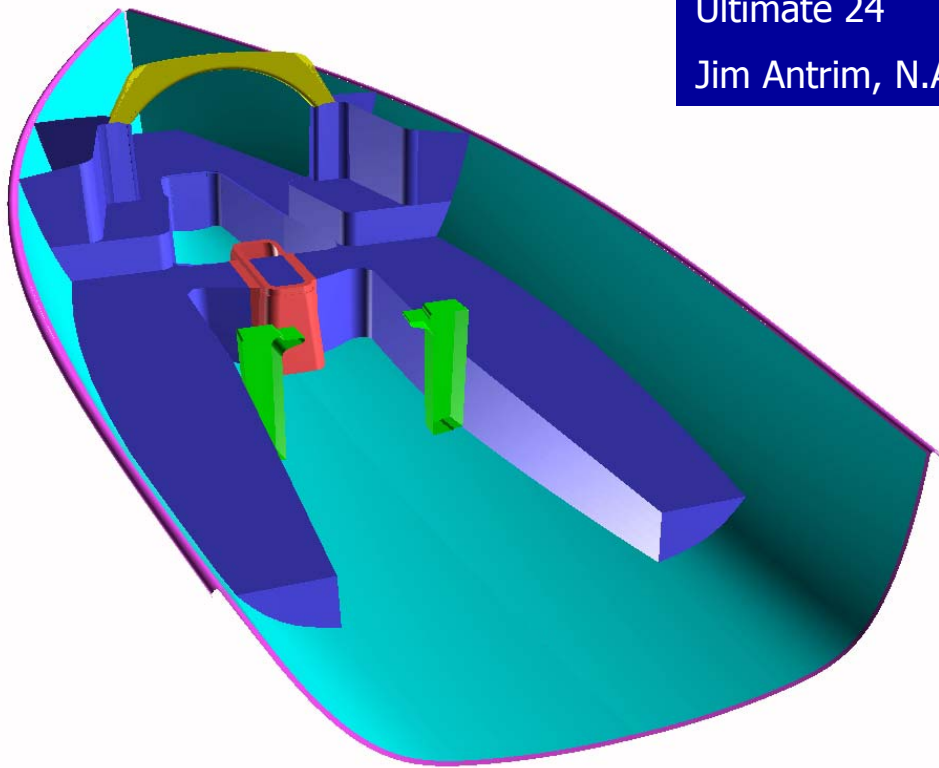
# Effective use of reinforcements in Ultimate 24 Sportboat



- Carbon uni strap aligned with primary loads: chainplate to keel, etc.
- Kevlar grounding patch in bottom
- Extra E-glass +/-45 biaxial in bow for collision and shear due to rig loads



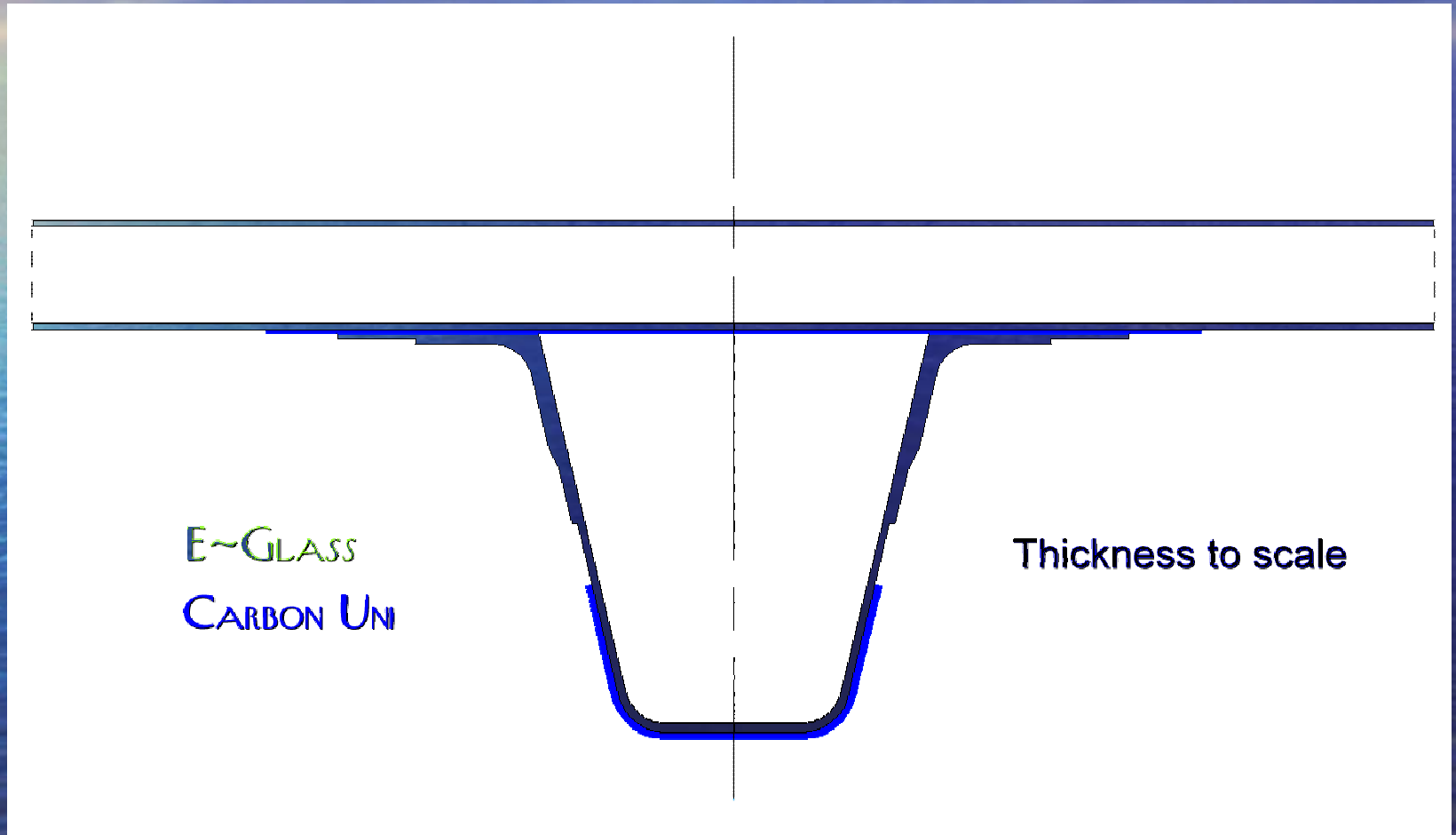
Ultimate 24  
Jim Antrim, N.A.



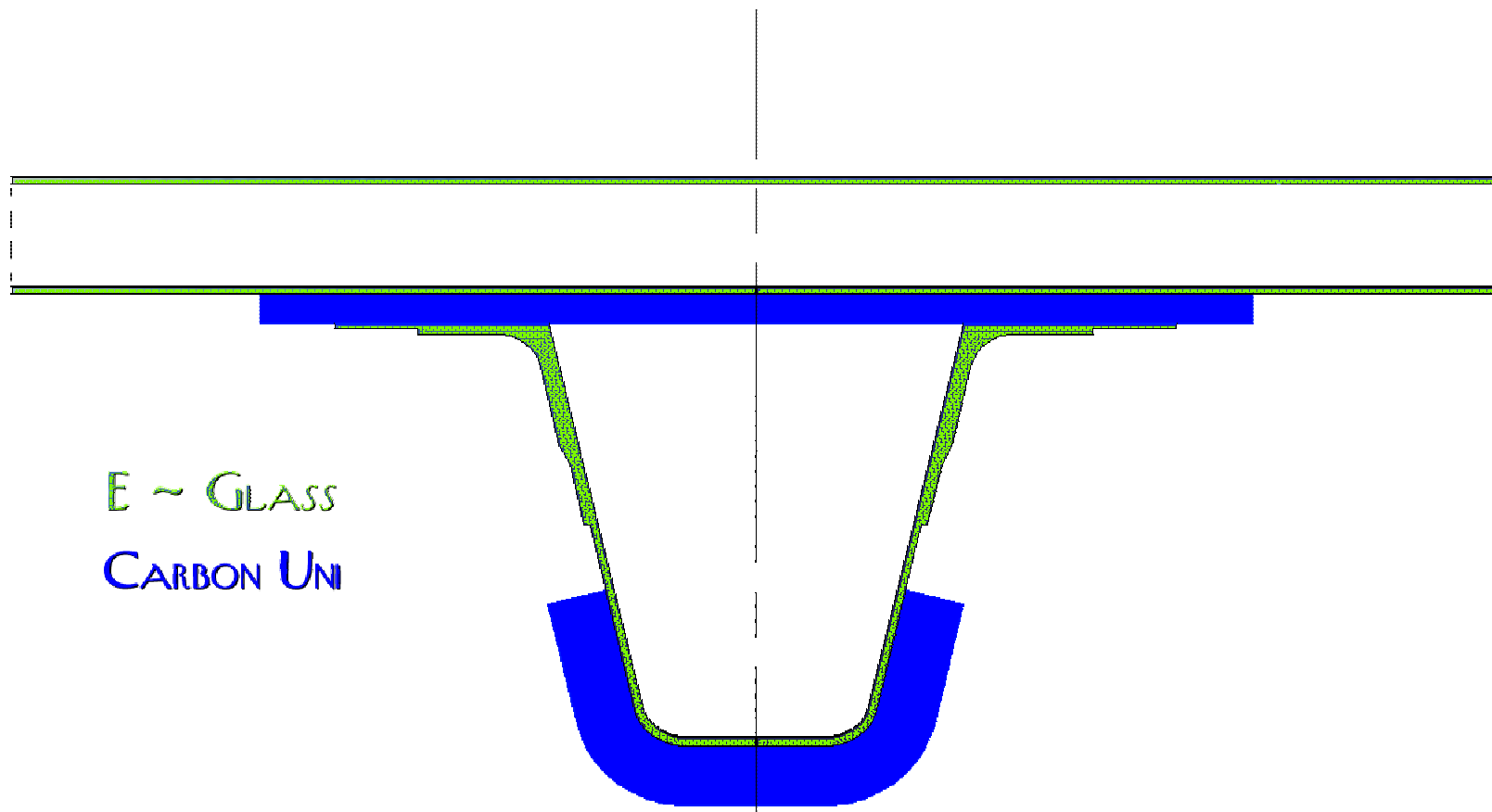


# Effective use of high tech composites

## Hat Section example

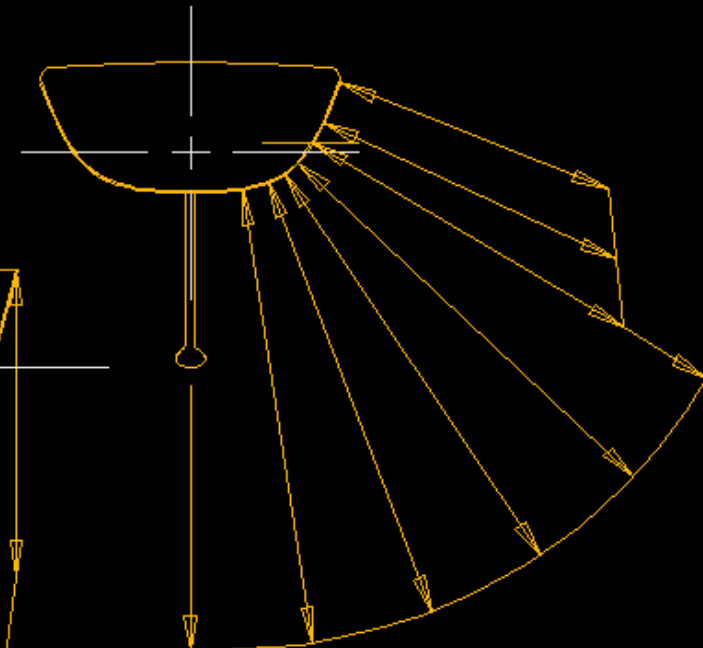
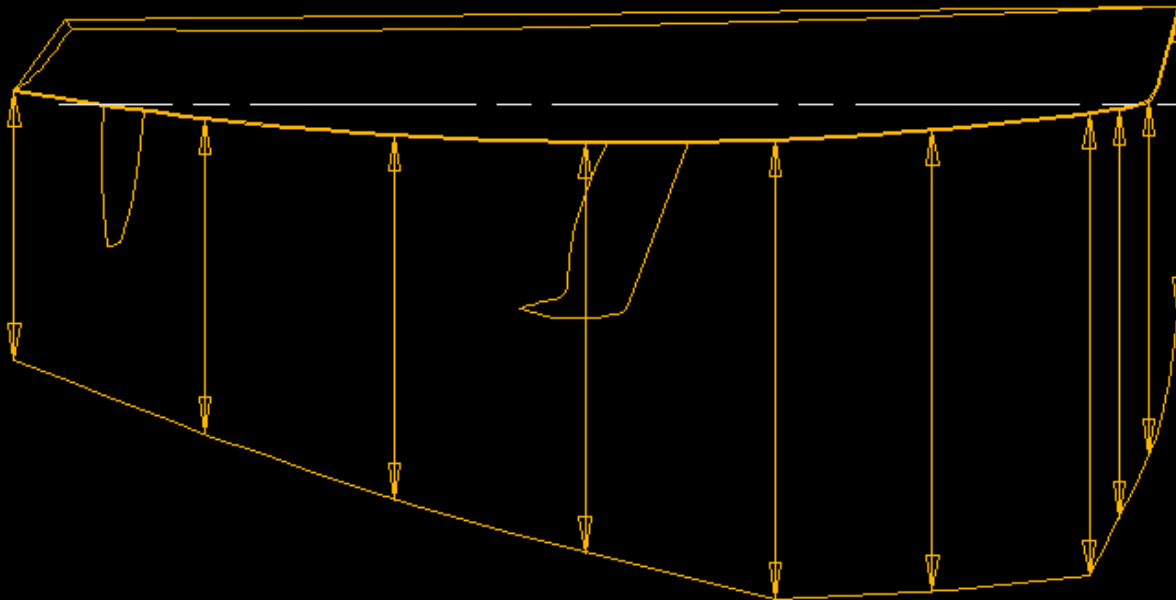


Thickness adjusted to EFFECTIVE area  
(Area times Young's Modulus)



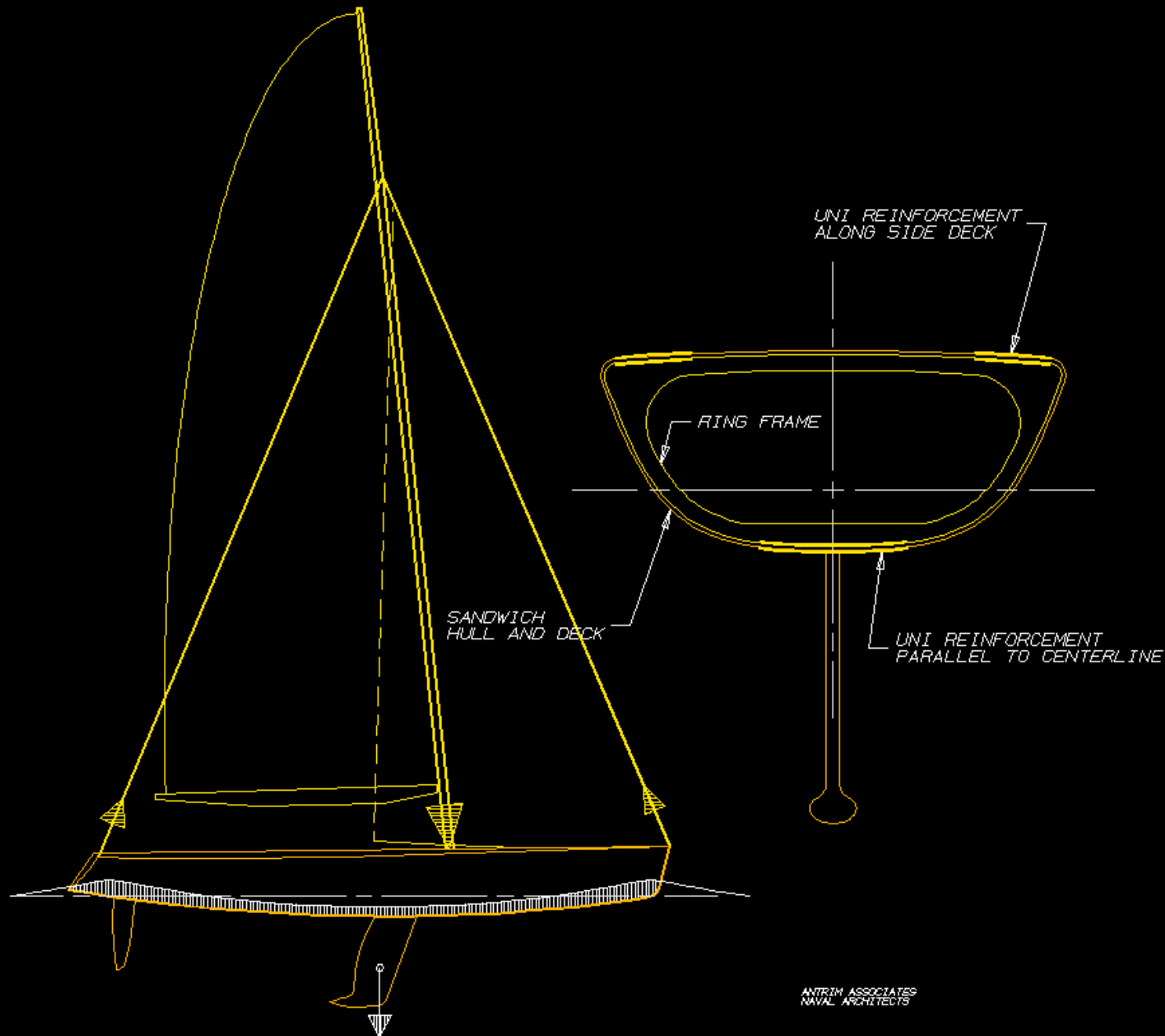
E ~ GLASS  
CARBON UNI

# Water head (pressure) on a monohull sailboat



Antrim Associates, Naval Architects  
[www.AntrimDesign.com](http://www.AntrimDesign.com)

Hull bending  
in a sailboat  
under  
imposed rig  
loads, wave  
loads, &  
keel weight



HULL LONGITUDINAL BENDING MOMENT  
INDUCED BY RIG TENSION AND WEIGHT/BUOYANCY DISTRIBUTION

ANTRIM ASSOCIATES  
NAVAL ARCHITECTS

*Perhaps we underestimated the hull bending loads?*



*Not* Antrim Associates, Naval Architects  
(though we have had mishaps on occasion too)

# Orienting fibers to the load

## Many parts have obvious load paths

- Tubes, such as mast, boom, bowsprit, rudder post, multihull crossbeams.
- Hull framing

## Hull/Deck shell has varied loading

- Water pressure, crew & gear loads perpendicular to panel surface
- Hull bending loads in plane with panel

## Bulkheads & ring frames have varied edge loading, shear



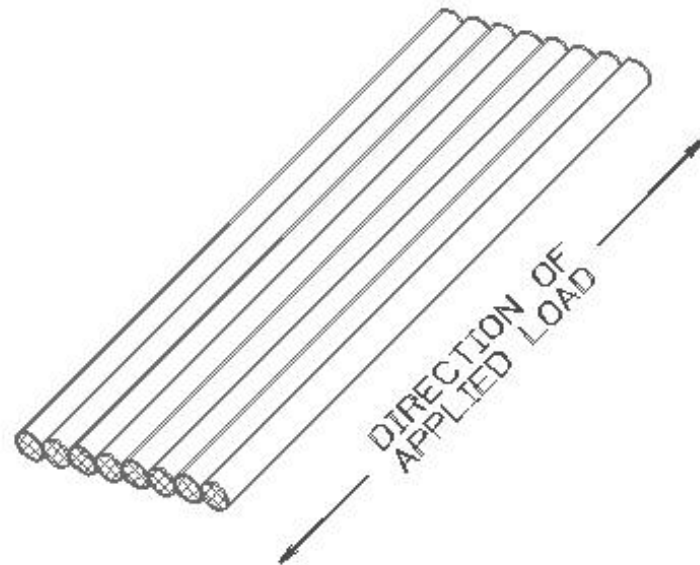
Billy Black photo

Antrim 40 trimaran *Zephyr*

# Ideal Composite Structure

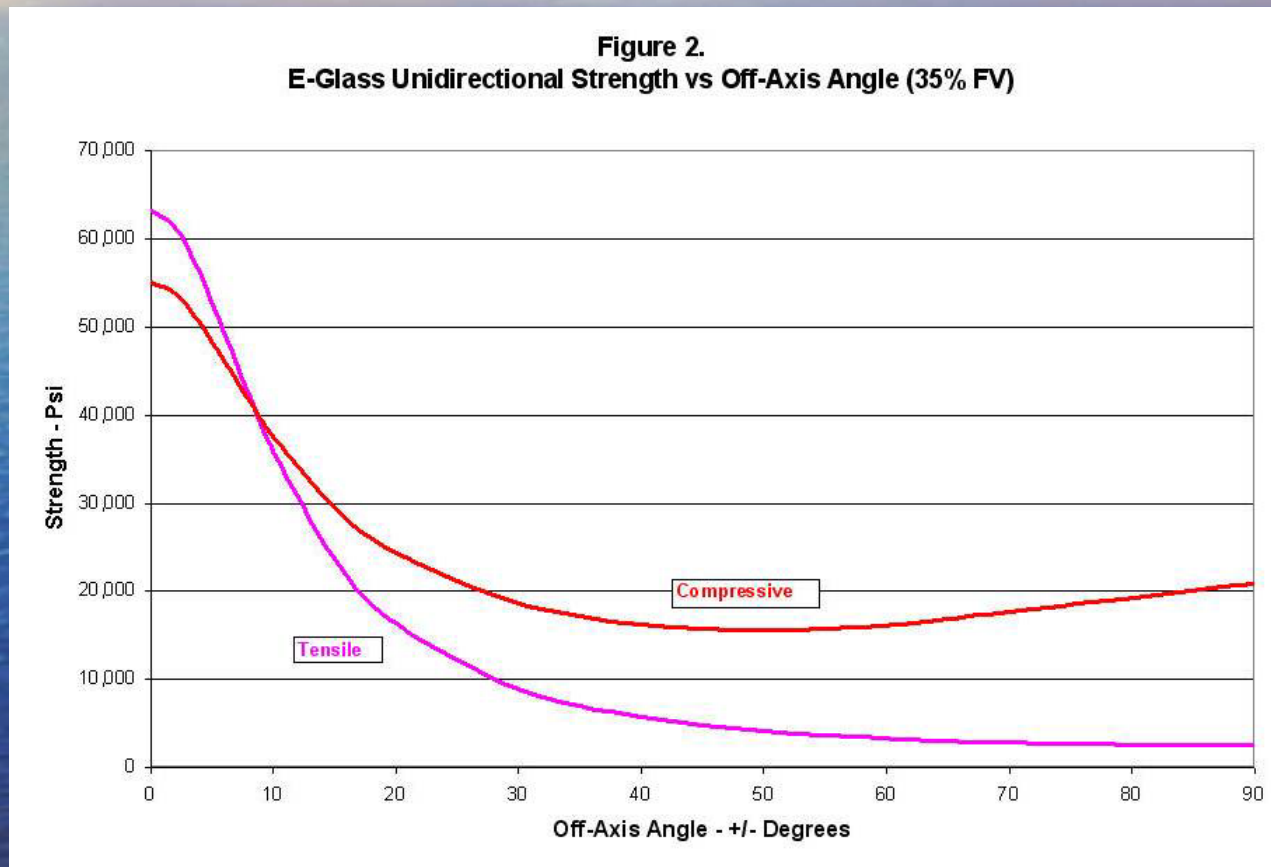
FIGURE 1

THE IDEAL COMPOSITE STRUCTURE



- \* ALL FIBERS RUN IN SAME DIRECTION
- \* MINIMUM RESIN ABOUT EACH FIBER
- \* NO CRIMPS, TWISTS, OR WAVINESS
- \* FIBERS ORIENTED IN DIRECTION OF LOAD

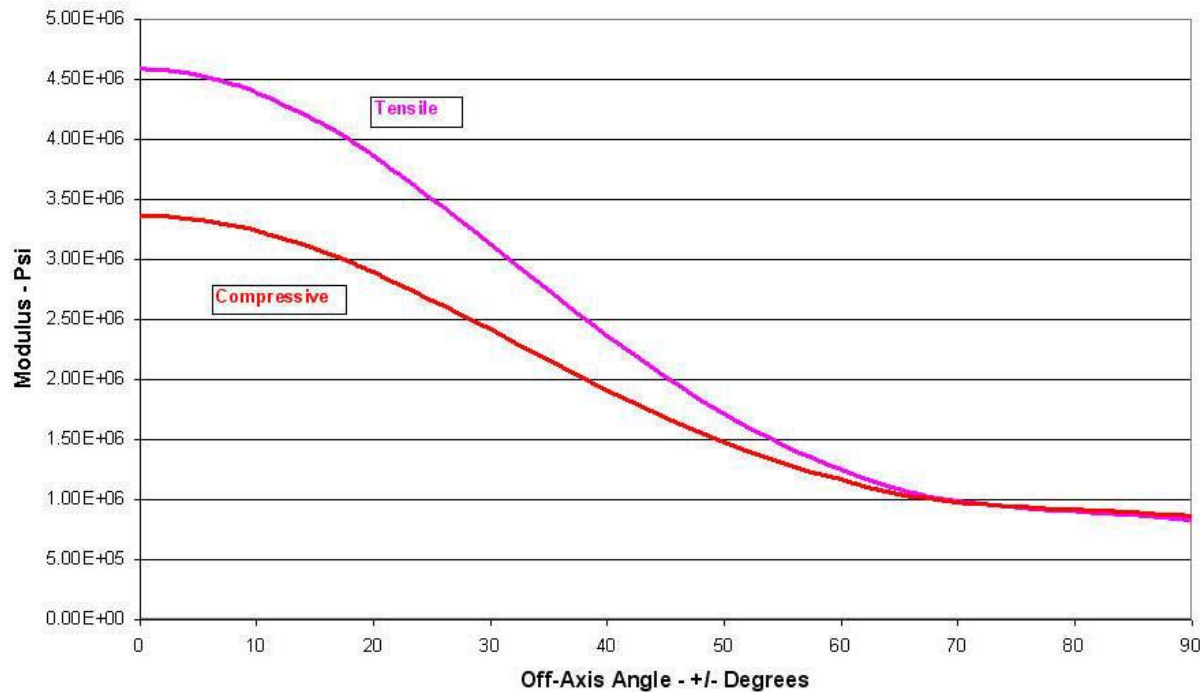
# E-Glass Unidirectional vs. Off-Axis Strength





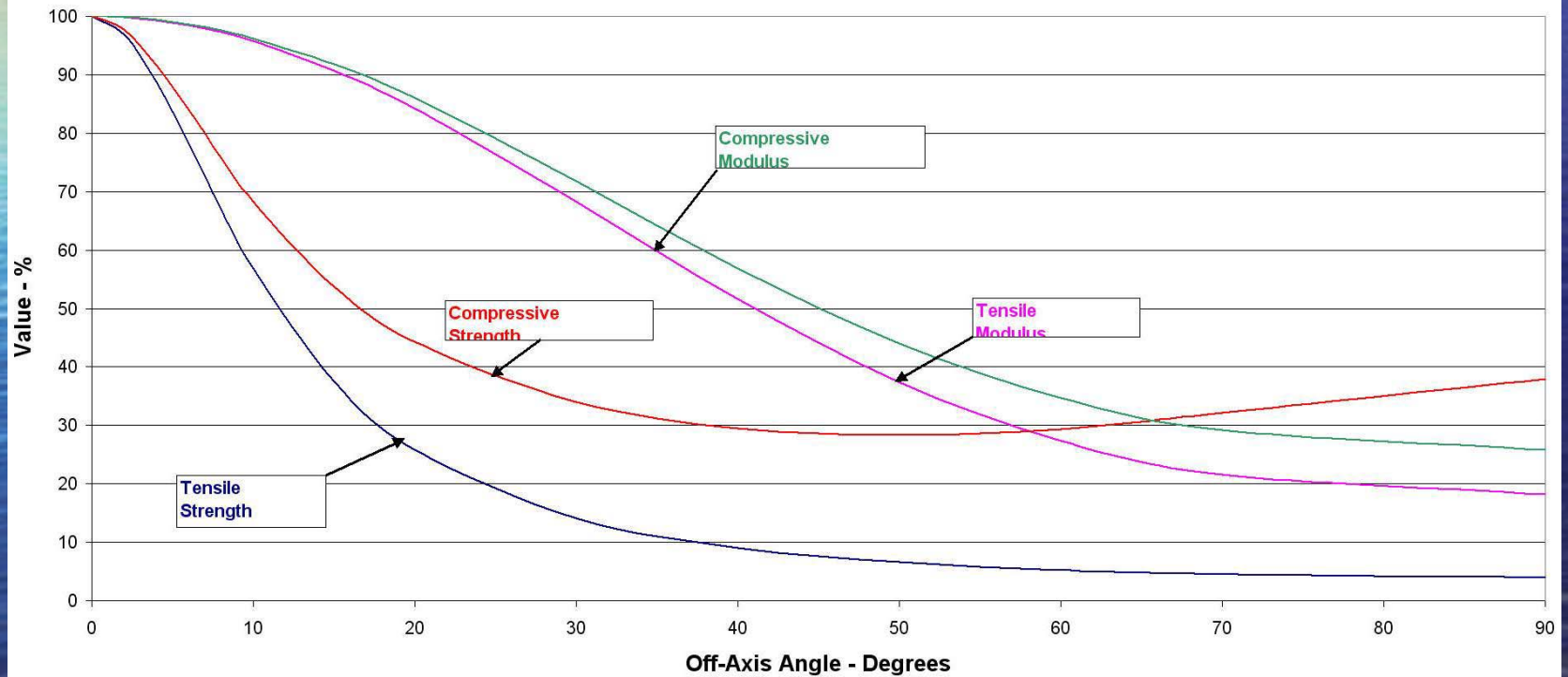
# E-Glass Unidirectional vs. Off-Axis Modulus

**Figure 3.**  
**E-Glass Unidirectional Modulus vs Off-Axis Angle (35% FV)**



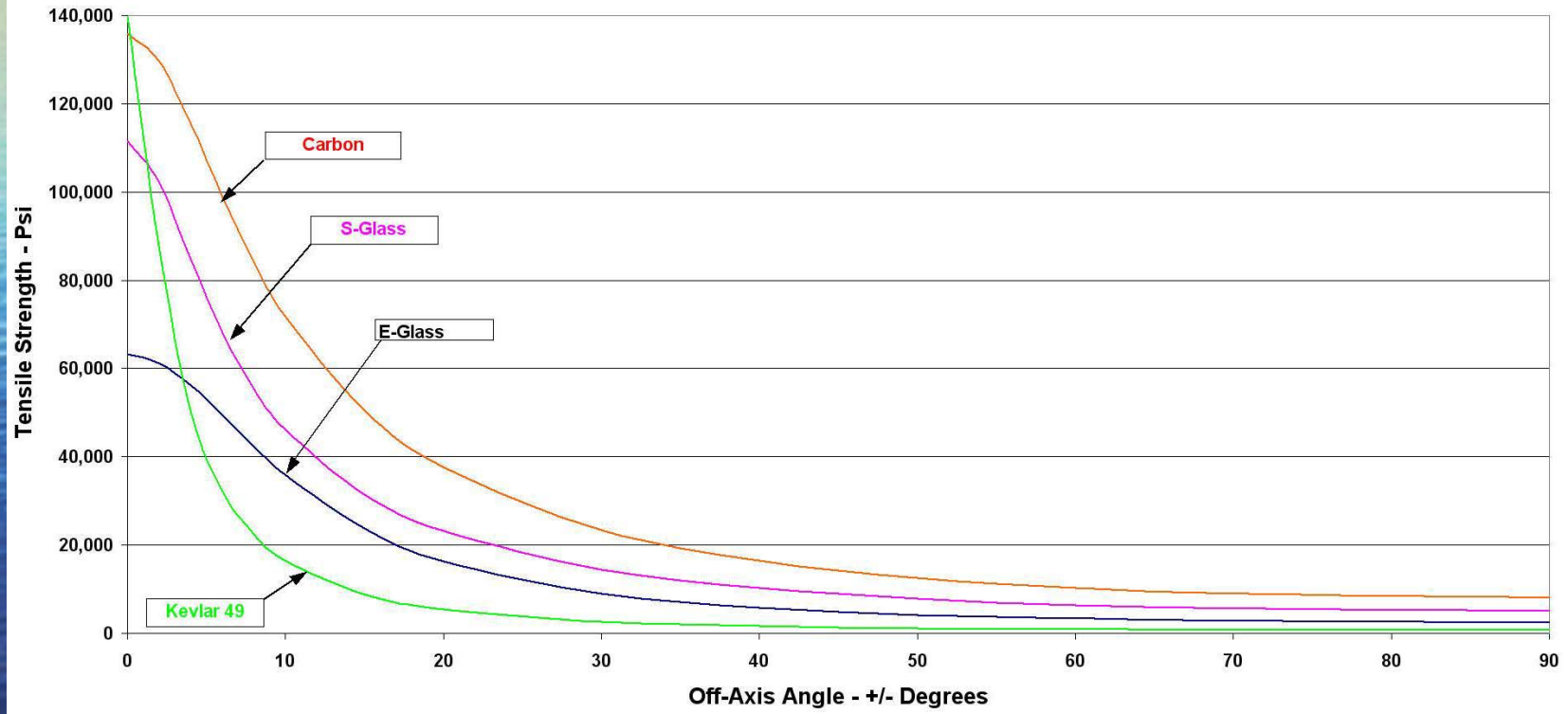
# E-Glass Unidirectional Mechanical Property % vs. Off Axis Angle

Figure 4.  
E-Glass Unidirectional Mechanical Property Percentages  
vs Off-Axis Angle (35% FV)



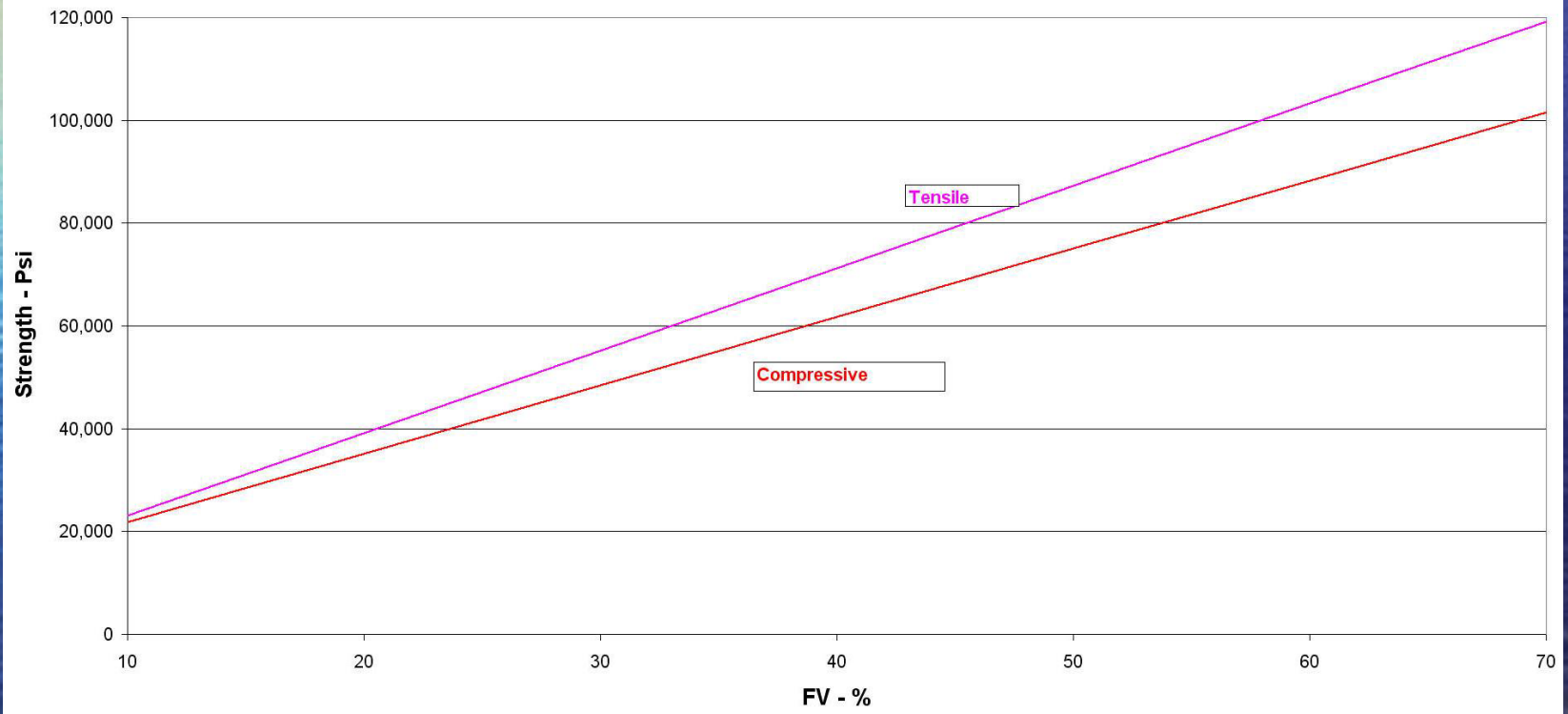
# Unidirectional Tensile Strength vs. Off Axis Angle

Figure 5.  
Unidirectional Tensile Strength vs Off-Axis Angle (35 % FV)



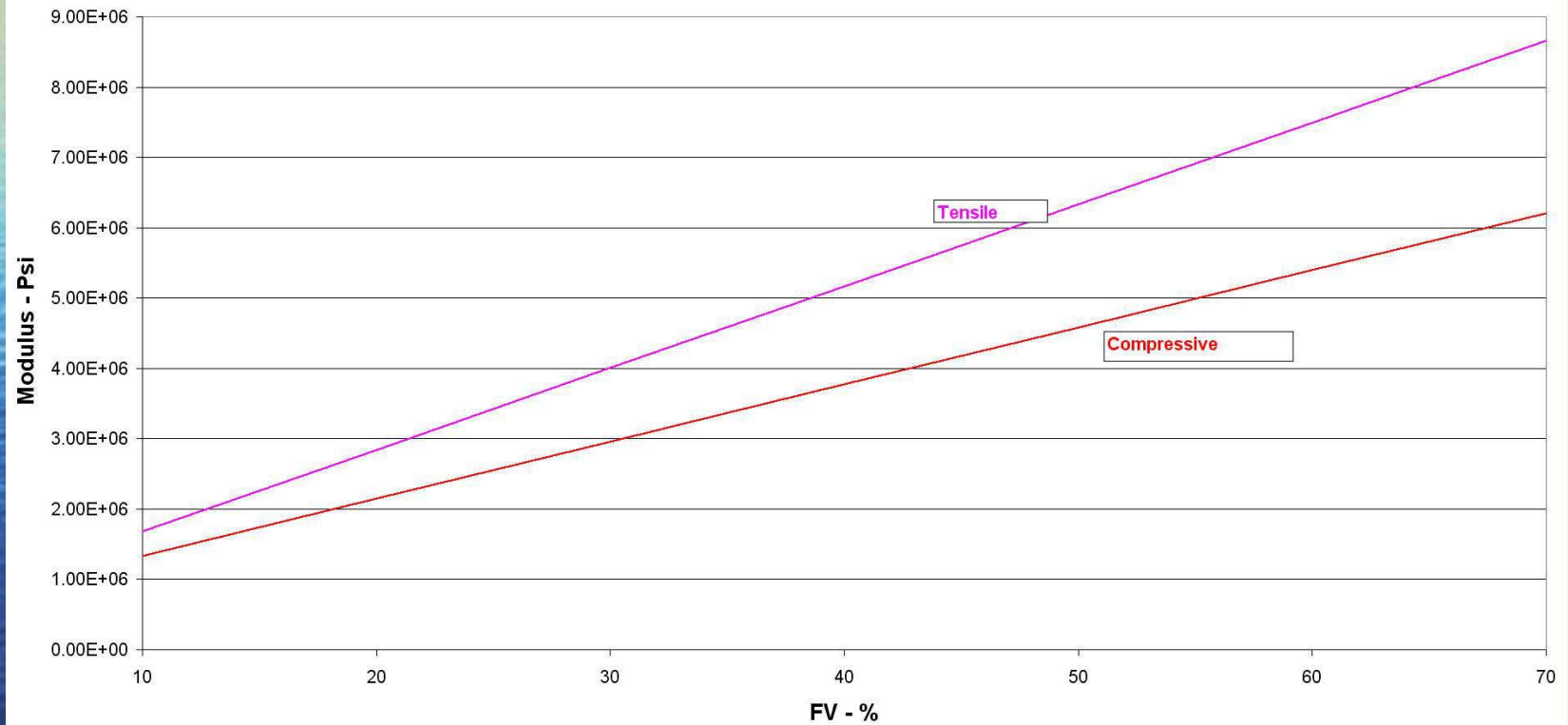
# E-Glass Unidirectional Strength vs. Fiber Volume

Figure 6.  
Unidirectional E-Glass Strength vs FV



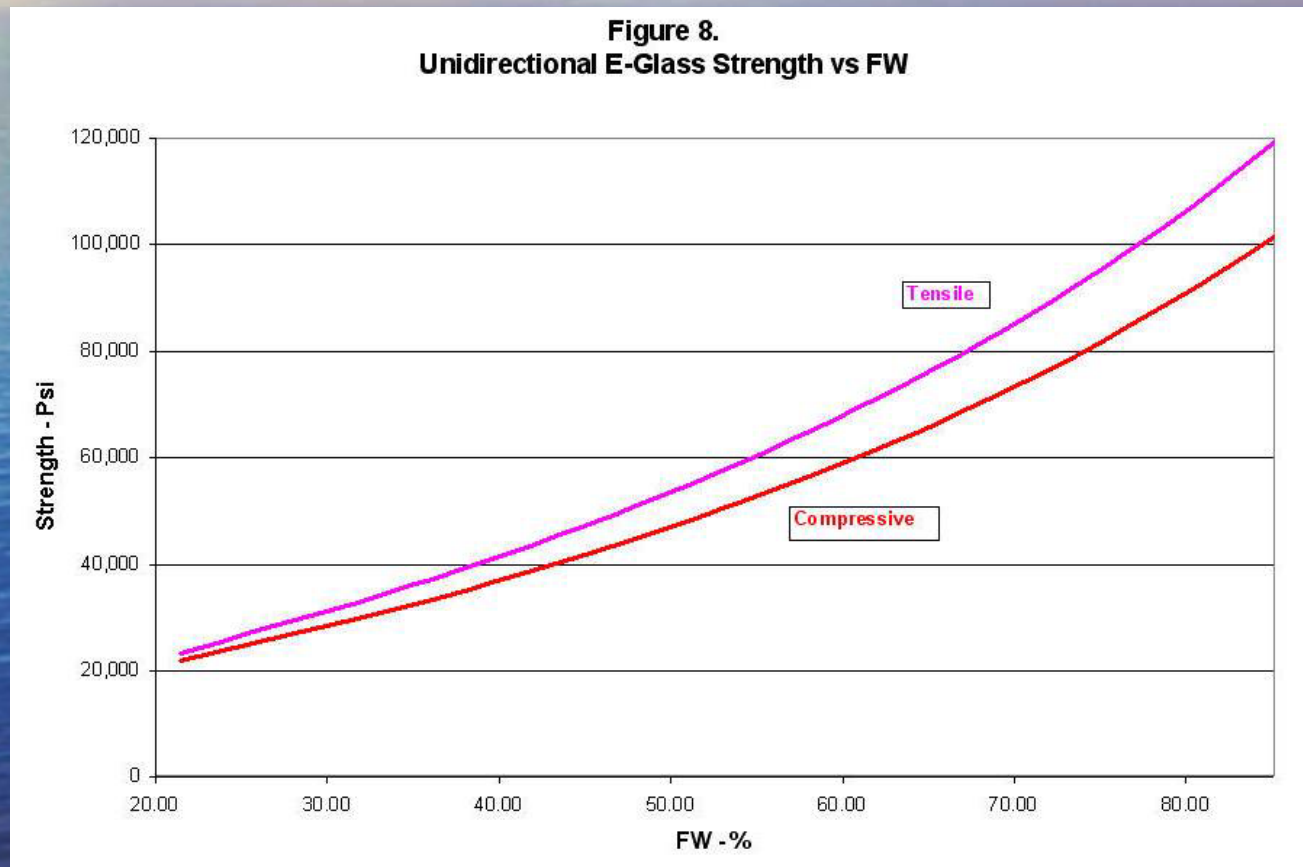
# E-Glass Unidirectional Modulus vs Fiber Volume

Figure 7  
Unidirectional E-Glass Modulus vs FV



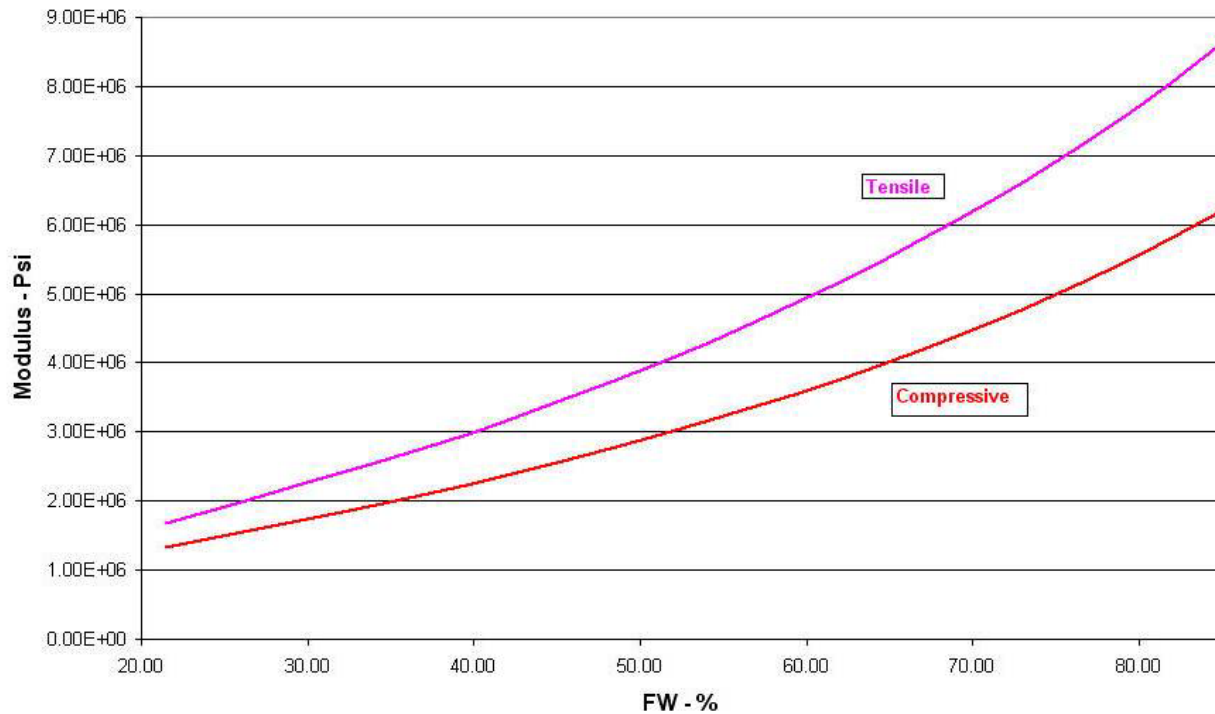
# E-Glass Unidirectional Strength vs. FW

Figure 8.  
Unidirectional E-Glass Strength vs FW

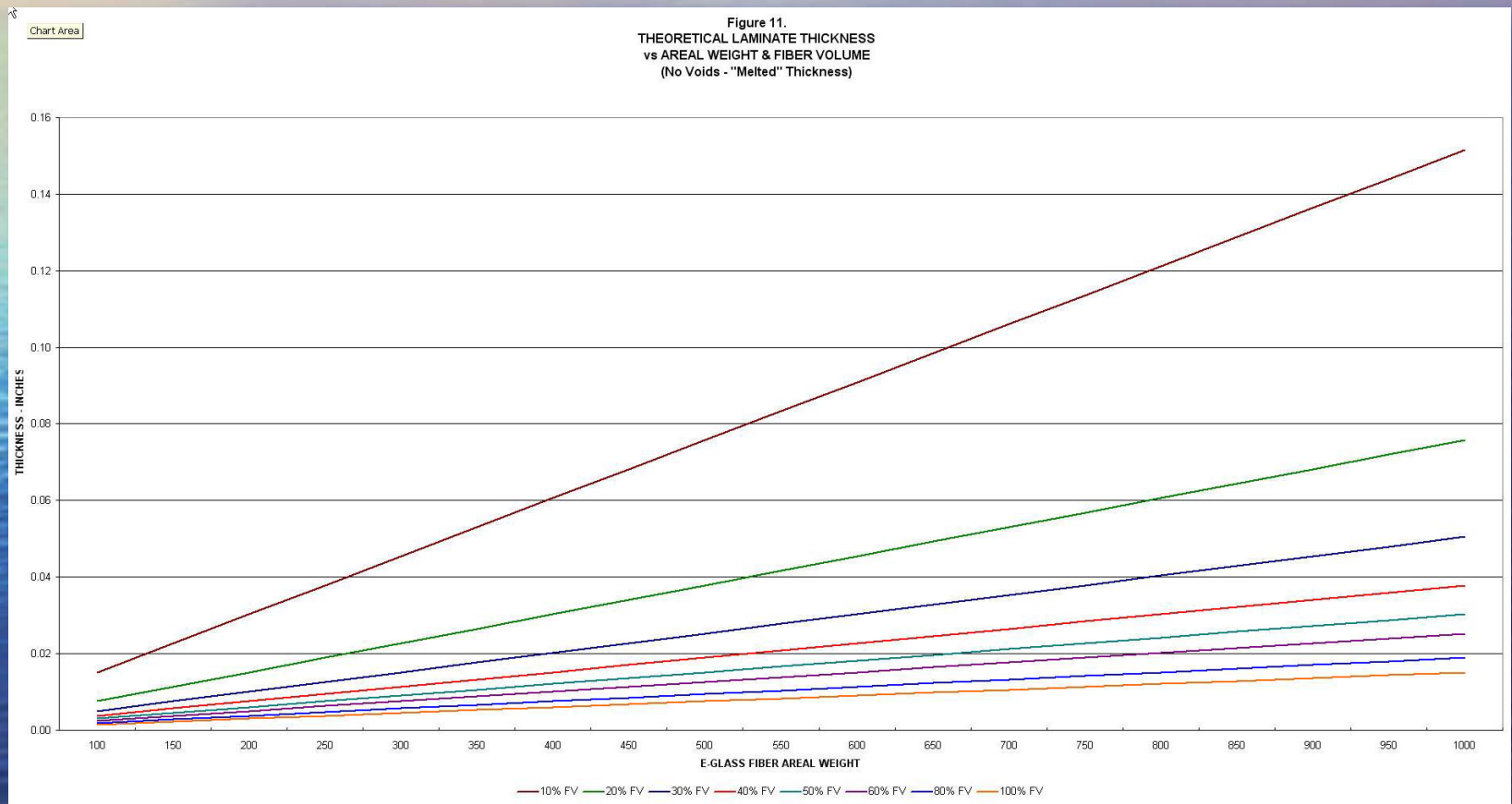


# Unidirectional E-Glass Modulus vs FW

Figure 9 .  
Unidirectional E-Glass Modulus vs FW



# Theoretical Laminate Thickness

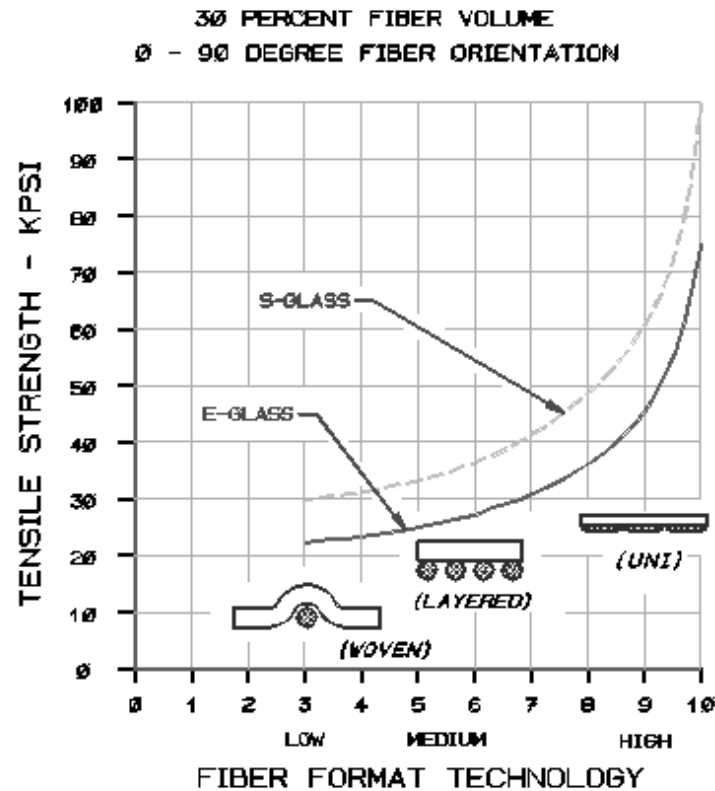




# Variation of Tensile Strength with Fiber Format

FIGURE 12

## VARIATION OF TENSILE STRENGTH WITH FIBER FORMAT



# Knockdown Factors for Woven Fabrics

**Figure 13.**

## Knockdown Factors for Woven Fabrics

E-Glass FV (%) =	25.00
Uni Strength (Psi) =	47,203
Uni E-Glass FV (%) =	25.00

Weave	Knockdown Factor (K)
Plain	0.80
4 HS	0.88
8 HS	0.49
Twill	0.70
Leno	0.73
Crowfoot	0.46
<b>Overall</b>	<b>0.75</b>

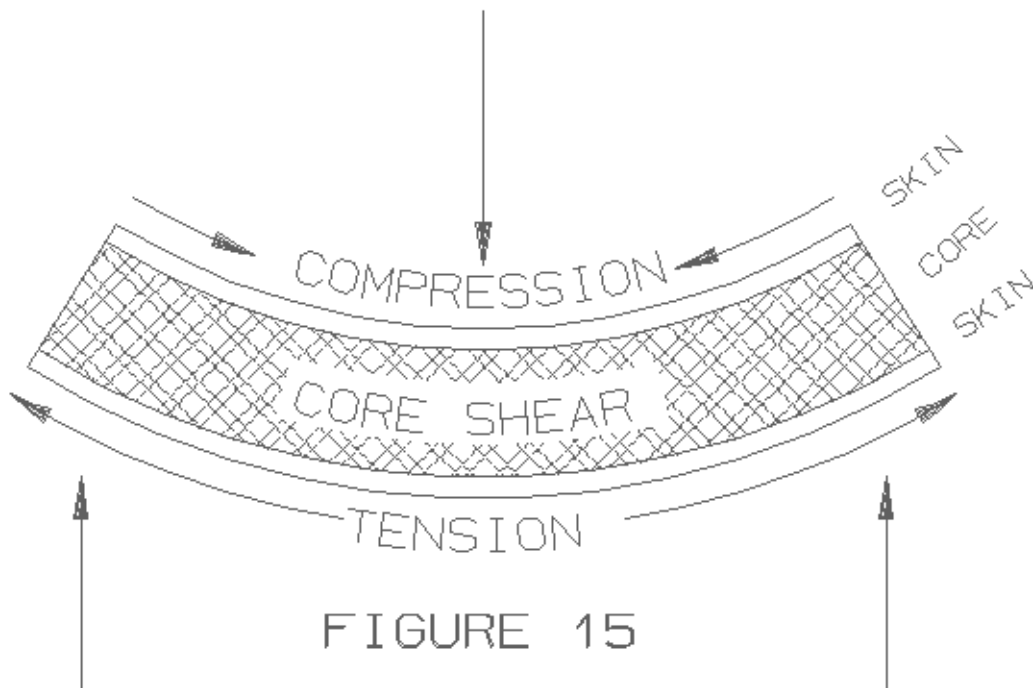
**Figure 14**

## Knockdown Factors for Woven Fabrics

E-Glass FV (%) =	25.00
Uni Strength (Psi) =	55,210
Uni E-Glass FV (%) =	30.00

Weave	Knockdown Factor (K)
Plain	0.69
4 HS	0.75
8 HS	0.42
Twill	0.60
Leno	0.62
Crowfoot	0.39
<b>Overall</b>	<b>0.64</b>

# Sandwich Laminate under Load



# Determining Composite Density & Thickness

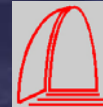
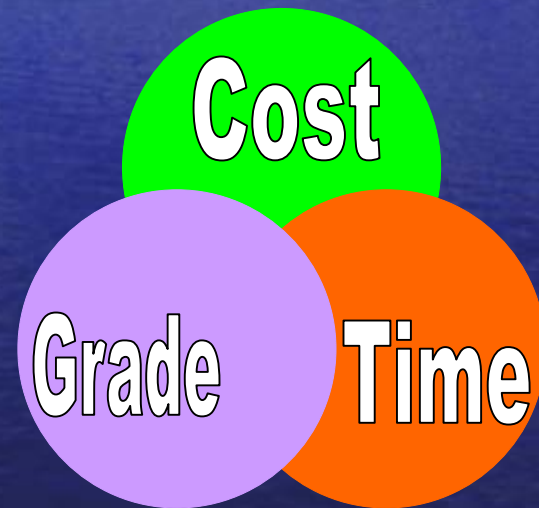
<b>Table 1</b>			
<b>Methods for Determining Composite Density &amp; Thickness</b>			
(From ASTM Composites Handbook Volume 1 Page 509)			
<b>GIVEN</b>		<b>UNKNOWN</b>	<b>EQUATION</b>
<b>% Fiber by Volume Fraction</b>			
Fiber Volume Fraction =	Fv	% Fiber by Weight =	Fv Fd
Fiber SpG =	Fd	(Fw)	Composite SpG
Resin Spg =	Rd	% Resin by Weight =	(1-Fv) Rd
Resin Volume Fraction =	1-Fv	(Rw)	Composite SpG
Resin Weight Fraction =	(1-Fv) Rd		
		Composite Spg =	Fv Fd + (1-Fv) Rd
<b>% Resin by Weight Fraction</b>			
Fiber Spg =	Fd	% Fiber by Volume =	(1-Rw) Composite SpG
Fiber Weight Fraction =	1-Rw	(Fv)	Fd
		% Resin by Volume =	Rw Composite SpG
Fiber Volume Fraction =	(1-Rw) Fd	(Rv)	Rd
Resin Spg =	Rd		
		Composite Spg =	1
			(1-Rw) / Fd + Rw / Rd
Resin Weight Fraction =	Rw		
Resin Volume Fraction =	Rw / Rd		
<b>Laminate Thickness</b>			
(No Voids - "Melted Thickness")			
Thickness t (Inches) =	<b>Total Laminate Weight (# / Sq.Ft.)</b>		
	Composite Spg x 5.2		
Total Laminate Weight =	Fiber Weight (# / Sq.Ft.)		
(# / Sq.Ft.)	Fw		

# E-Glass Composite Density & Thickness

<b>Table 2.</b>			
<b>E-Glass Composite Density &amp; Thickness</b>			
<i>GIVEN</i>		<i>UNKNOWN</i>	
<b>Resin by Weight Fraction</b>			
Fiber Volume Fraction	0.350	% Fiber by Weight	53.846
Fiber Density (Spg)	2.600		
Fiber Weight Fraction	0.910		
Fiber Weight (# / Cu.In.)	0.033		
Resin Density (Spg)	1.200	% Resin by Weight	46.154
Resin Volume Fraction	0.650		
Resin Weight Fraction	0.780		
Resin Weight (# / Cu.In.)	0.028		
		Composite Density (Spg)	1.690
		(# / Cubic Inch)	0.0610
<b>Resin by Weight Fraction</b>			
		% Fiber by Volume	35.000
Fiber Density (Spg)	2.600		
Fiber Weight Fraction	0.538		
Fiber Weight (# / Cu.In.)	0.014		
Fiber Volume Fraction	0.150		
		% Resin by Volume	65.000
		Composite Density (Spg)	1.690
		(# / Cubic Inch)	0.0610
Resin Density (Spg)	1.200		
Resin Weight Fraction	0.462		
Resin Weight (# / Cu.In.)	0.037		
Resin Volume Fraction	0.850		
<b>Laminate Thickness</b>			
(No Voids - "Melted Thickness")			
Fiber Areal Weight (Grams / Meter <sup>2</sup> )	300		
Fiber Weight (Oz./Yd. <sup>2</sup> )	8.850		
Fiber Weight (# / Sq.Ft.)	0.0615		
Resin Weight (# / Sq.Ft.)	0.0527		
Laminate Weight (# / Sq.Ft.)	0.1141		
Thickness t (Inches) =	0.0130		

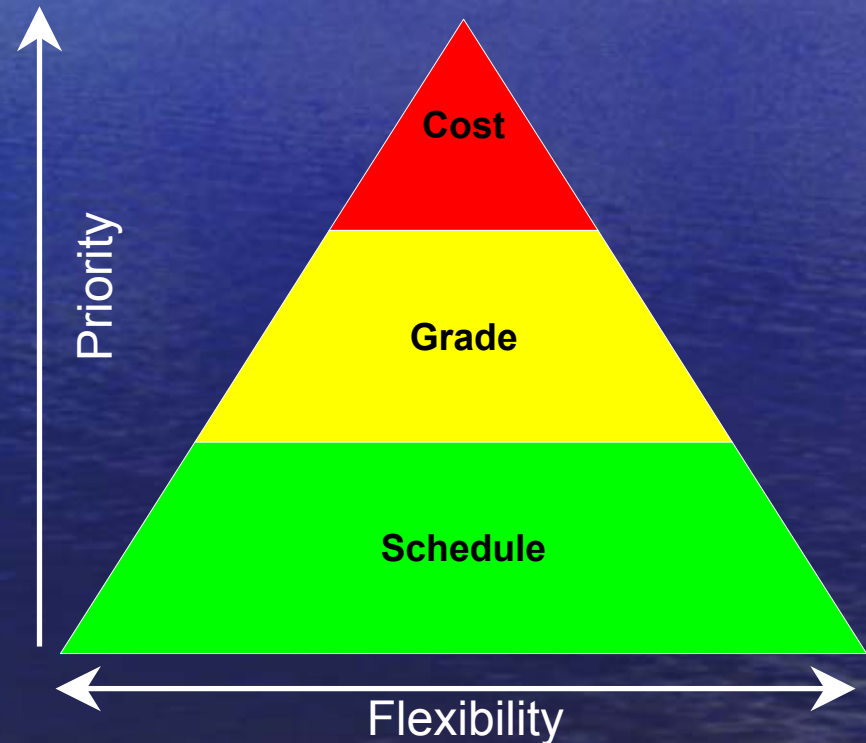
# Triple Constraint

- Cost / Resources
  - Amount of \$, materials or personnel to support the project
- Schedule
  - Amount of time to complete the project
- Quality / Grade / Performance
  - Quality = Fit for purpose
  - Grade = Degree of luxury
    - (Product) Performance – For yachts, this can be a sub-category of grade



# Client Priorities

- Client priorities will determine how tradeoffs should be prioritized
  - If Cost is the primary driver (fixed) there should be some flexibility to “grade” and / or schedule
  - If Schedule is the driver, there may be increased cost and compromises on “grade”
  - If Quality / Grade is the driver, there may be increased cost and compromises on schedule



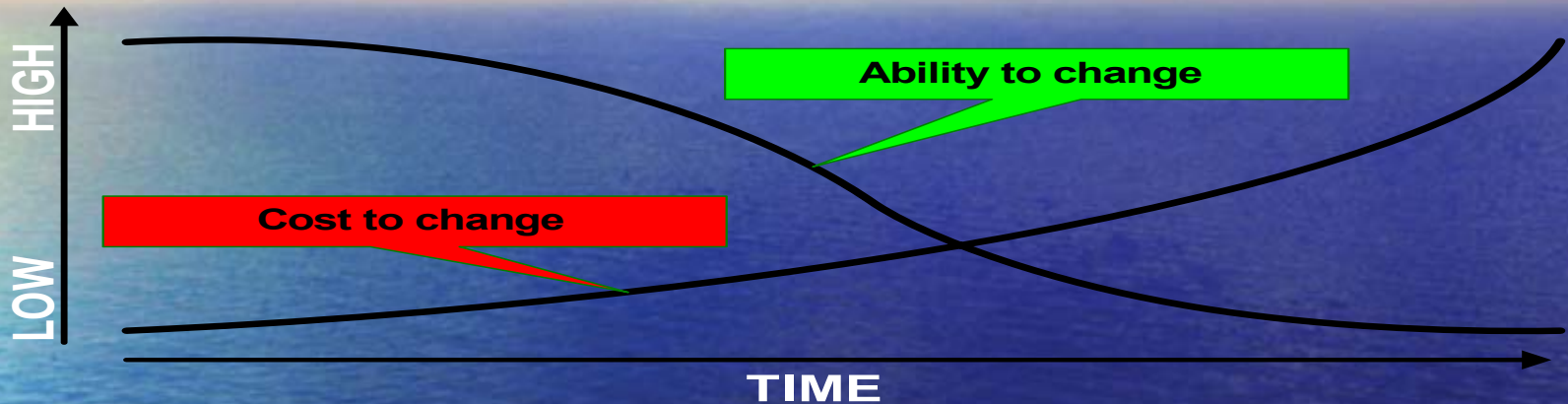
# Applying Priorities

- If Cost is the primary driver
  - Exotic or expensive materials can be avoided
    - i.e. E-Glass instead of Carbon would be a compromise on “grade”
- If Schedule is the primary driver
  - You may invest additional resources (people or \$) to finish the job quicker
- If Grade is the driver
  - It may take longer to get the materials (teak decking?)
  - It will cost more than a comparable vessel without the “options”





# Planning ahead



- By determining what the primary driver is during the early phases of the design process, costly and time consuming changes can be avoided later
- Every project follows a lifecycle
  - Initiating → **Planning** → Executing → Monitor/Control → Close<sup>1</sup>
  - A clear plan will help the client, designer and builder understand each other's perspective, constraints and requirements

1. Project Management Institute PMBOK 3rd edition



# Estimating Schedule

- Expert Judgment
  - Work with the builder to estimate how long a project will take
    - Break the project into phases: tooling, hull, deck, joining, assembly, finishing, etc.
    - Take into account that if there are multiple projects or production lines, there will be interruptions!
- Analogous Estimating
  - If you have done similar projects previously:
    - Refer to how long each step took
    - Refer to the “lessons learned” from the project (you did write them down, right?)
- Parametric Analysis
  - Duration = Amount of Work x Productivity
    - i.e. Each ply takes 30 minutes and we have 6 plies to lay up...
- Three Point Estimate
  - $(\text{Pessimistic} + 4 \times \text{Most Likely} + \text{Optimistic}) / 6$



# Estimating Cost (and weight)

- Bottoms-Up Estimate
  - An accurate weight analysis provides:
    - A Bill of Materials (BOM) for constructing the boat
      - Each of the BOM components should detail both weight and cost
    - A listing of the “standard” equipment for the boat
      - This will determine “empty” displacement
      - Adding cost information will verify if you are within budget constraints
    - A listing of the “optional” as well as personal gear for a standard weekend
      - This will determine “light” displacement
    - A listing of what an owner would provision for a long trip
      - This will determine “loaded” displacement
- Analogous Estimate
  - What have similar projects cost in the past (+ inflation)?
- Parametric Estimating (good for hardware)
  - $\text{Cost} = \text{Component Cost} \times \text{Quantity}$
- Vendor Bid Analysis

